# Proposed test method to establish hydrogen compatibility of materials for fuel cell vehicles

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# Motivation: establish materials compatibility for high-pressure hydrogen service in context of hydrogen fuel cell electric vehicles

#### **Goals of presentation:**

- Briefly summarize activity within SAE Fuel Cell Safety Task Force
  - SAE H2 Compatibility Expert Team
  - Collaborative testing and test criteria development
- Present test criteria developed for SAE J2579
  - Brief justification of requirements for materials compatibility

### **SAE Fuel Cell Safety Task Force**

- Meets quarterly with broad representation from automotive OEMs
- Tasked with developing several standards for fuel cell vehicles in context of safety
  - J2579 Standard for Fuel Systems in Fuel Cell and Other Hydrogen Vehicles
    - Includes requirements for materials in contact with high-pressure gaseous hydrogen
  - SAE H2 Compatibility Expert Team
    - Formed to develop requirements for hydrogen compatibility of materials
    - Includes hydrogen compatibility experts identified by Task Force representation

### **SAE H2 Compatibility Expert Team**

- Representation from nationally funded research programs funded to enable deployment of fuel technologies
  - Germany: MPA Stuttgart
  - Japan: Kyushu University and AIST
  - US: Sandia National Laboratories
- Collective learning through so-called "round robin" testing campaign
  - Development of capabilities and examination of procedures to execute fatigue tests in high-pressure hydrogen at low temperature
  - Demonstrate test methodologies at MPA, KU and SNL

### Collective learning activity ("round robin")

Test	Test conditions	Environment	Number of tests
Slow strain rate tension (SSRT)	≤ 5 x10 <sup>-5</sup> s <sup>-1</sup>	Control -40°C	3
	≥ 5 X10°S'	90 MPa H2 -40°C	3
Notched	Sa = 200 MPa	Control -40°C	3
tension-tension fatigue	R = 0.1 1 Hz	90 MPa H2 -40°C	3
Smooth Sa tension-compression fatigue	Sa = 320 MPa R = -1	Control -40°C	3
	1 Hz	90 MPa H2 -40°C	3

## Test criteria for hydrogen compatibility of materials

SAE J2579, Appendix B.3 is essentially a set of generic test criteria for evaluation of structural metals for service in high-pressure gaseous hydrogen

- Part 1: Definition of materials and environment al test conditions
- Part 2: SSRT
- Part 3: Fatigue life test
- Part 4: Welds

In general, CSA CHMC1 is referenced for the test methods (CHMC1 references ASTM standards)

### Part 1: Definition of materials and environmental test conditions

- Material must be defined by and satisfy requirements for
  - Composition
  - Tensile properties: specified minimum Sy, Su, El
- Environmental test conditions
  - Pressure ≥ 1.25 NWP (nominal working pressure)
  - Test temperature: 228 K (for most materials)
  - Measured gas purity according to CSA CHMC1
    - 2 ppm O<sub>2</sub>, 10 ppm H<sub>2</sub>O

# Part 1: Definition of materials and environmental test conditions: test temperature

**Table B.3.1.4 from SAE J2579** 

Alloy type	Test method	Test temperature (K)		
Austenitic stainless steel	SSRT	228 ±5		
	Fatigue life	228 ±5 and 293 ±5		
Nickel-based alloys	SSRT and Fatigue life	228 ±5		
Aluminum, magnesium and copper alloys	SSRT and Fatigue life	293 ±5		
Other alloys	SSRT and Fatigue life	228 ±5 and 293 ±5		

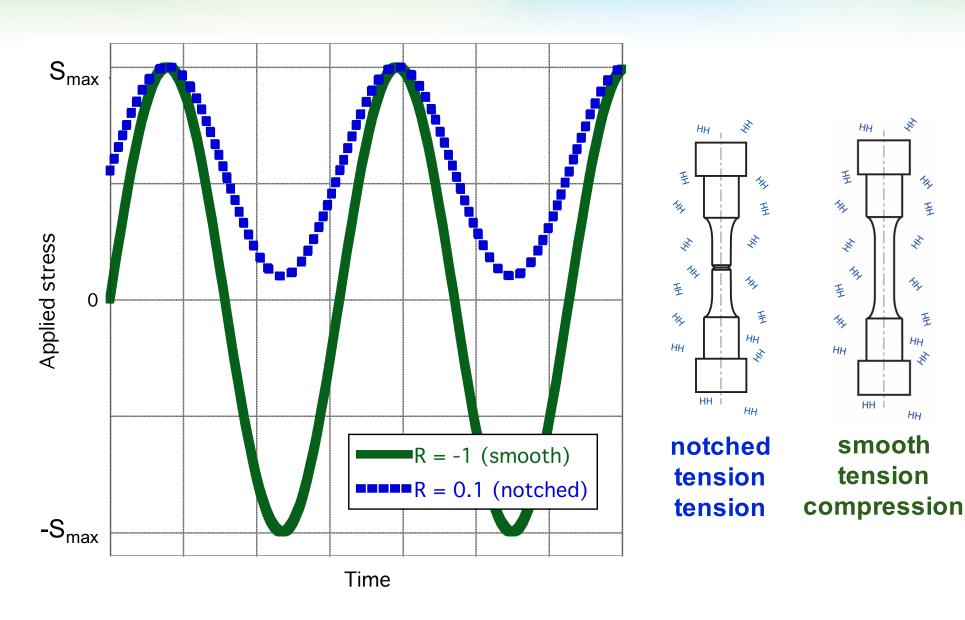
#### Part 2: Slow strain rate tension test

- Basic tensile test at slow strain rate in the defined hydrogen environment
- Minimum of three (3) tests
- Average property values must be greater than the specified minimum Sy and Su values respectively
- Average elongation (EI) > 12%
- Additionally, Su/Sy > 1.07

### Part 3: Fatigue life test

- Force-controlled (axially loaded cylindrical) fatigue test in the defined hydrogen environment
  - Frequency of 1 Hz
  - Maximum stress shall be 1/3 of measured Su (air)
- Minimum of three (3) tests
- Two test configuration options
  - Option 1: smooth test specimen with R = -1, or
  - Option 2: notched test specimen with R = 0.1
- Cycles to failure >200,000 cycles for each test
  - Alternatively, cycles to failure >100,000 cycles for each of 5 notched test specimens

### Part 3: Fatigue life test: stress cycle



1/4

нн і

smooth

tension

#### Part 4: Welds

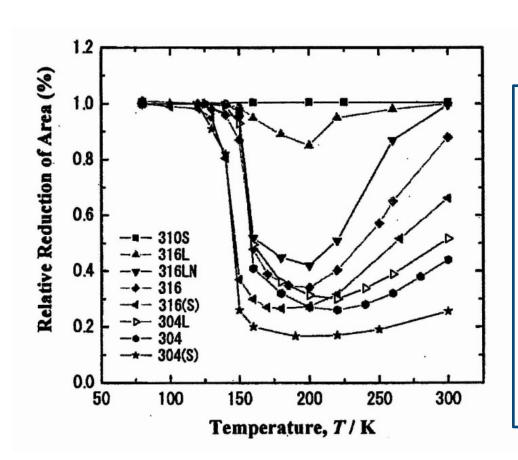
- Prepare representative welded structures
- Same testing requirements as for non-welded materials
  - Specified minimum tensile properties must be defined
  - Weld material must satisfy the minimum specified properties
  - Average values from SSRT tests of weld-material must satisfy minimum specified strength properties, El > 12% and Su/Sy > 1.07
  - Fatigue life must be >200,000 cycles for each of three (3) smooth or notched fatigue tests; or >100,000 cycles for each of (5) notched fatigue tests

### Summary of requirements for compatibility

Test configuration		Evaluation parameter	Requirements of tests performed in H2		
Slow strain rate tension tests – SSRT (3 tests)		Yield strength	Average ≥ Sy		
		Tensile strength	Average ≥ Su		
		Strain hardening capacity	Average > 1.07		
		Elongation	Average ≥ 12%		
Fatigue life tests (must satisfy 1 of 3 options)	Option 1 (3 tests): Smooth, R= -1	Cycles to failure	Each > 200,000 cycles		
	Option 2 (3 tests): Notched, R = 0.1	Cycles to failure	Each > 200,000 cycles		
	Option 3 (5 tests): Notched, R = 0.1	Cycles to failure	Each > 100,000 cycles		

Note: Sy and Su are specified minimum yield and tensile strength respectively

# Tensile properties are degraded in gaseous hydrogen especially at low temperature



#### Requirement:

- Minimum specified strength properties are maintained
- Ductility is consistent with pressure applications

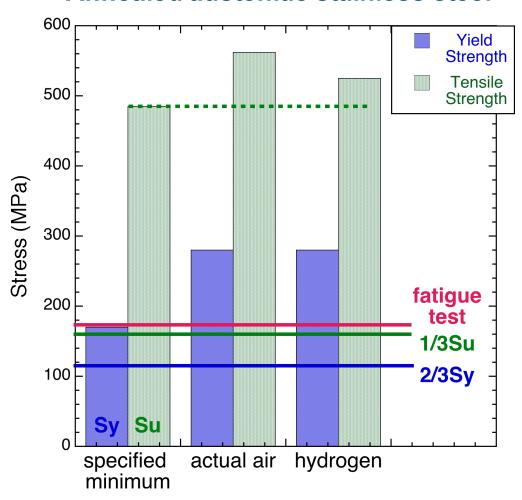
#### **Rationale:**

 Known and ductile tensile response

Data from: Fukuyama et al., *J Japan Inst Metals* 67 (2003) 456-459.

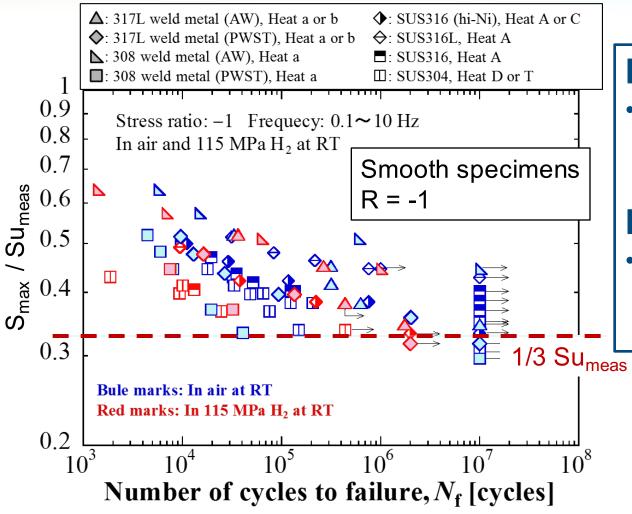
# Tensile strength properties are not degraded in gaseous hydrogen for acceptable materials

#### Annealed austenitic stainless steel



- Common stress limitations for fatigue design: minimum of 2/3 Sy and 1/3 Su
- Yield and tensile strengths are typically not affected by hydrogen
- Maximum stress during fatigue testing (J2579) always greater than 1/3 Su

# Fatigue life of smooth specimens is typically infinite at stress of 1/3 Su<sub>meas</sub>



#### Requirement:

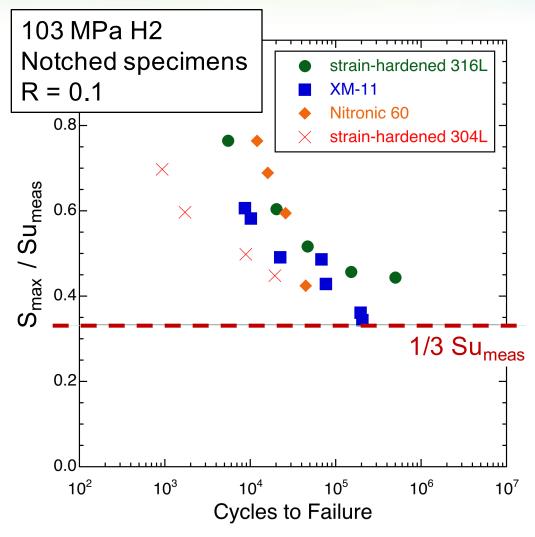
• Nf > 200,000 cycles at  $S_{max} = 1/3 Su_{meas}$ 

#### **Rationale:**

 Ensure fatigue life at high stress is >> than design life

Data from: M. Nakamura et al., M&M2017 conference, 7-9 October 2017, Hokkaido, Japan

# Notched specimens assess sensitivity to stress concentration for typical maximum stress (1/3Su)



#### **Requirement:**

• Nf > 100,000 cycles at  $S_{max} = 1/3 Su_{meas}$ 

#### Rationale:

 Ensure fatigue life at high stress is >> than design life

Data from: C. San Marchi et al., 43<sup>rd</sup> MPA Seminar, 11-12 October 2017, Stuttgart, Germany

# Diverse range of austenitic stainless steels have been evaluated, including high-strength alloys

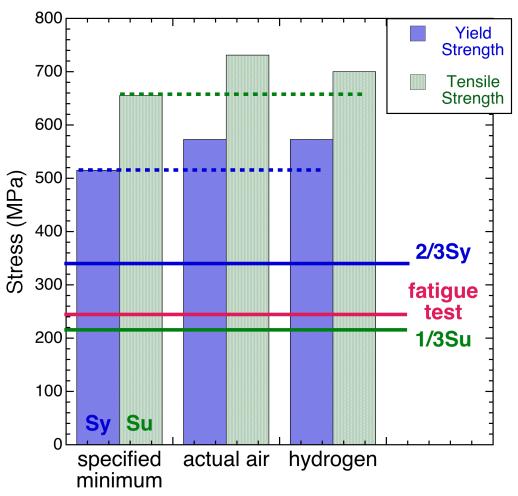
material	Yield (MPa)	Tensile (MPa)	Cr	Ni	Mn	N	Typical allowable stress (MPa)
316L	280	562	17.5	12	1.2	0.04	115
CW 316L	573	731	17.5	12	1.2	0.04	218
304L	497	721	18.3	8.2	1.8	0.56	195
XM-11	539	881	20.4	6.2	9.6	0.26	207
Nitronic 60	880	1018	16.6	8.3	8.0	0.16	218
SCF-260	1083	1175	19.1	3.3	17.4	0.64	333

Wide range of strength

Wide range of Ni/Mn content

# High-strength materials can be evaluated by method and enable higher stress designs

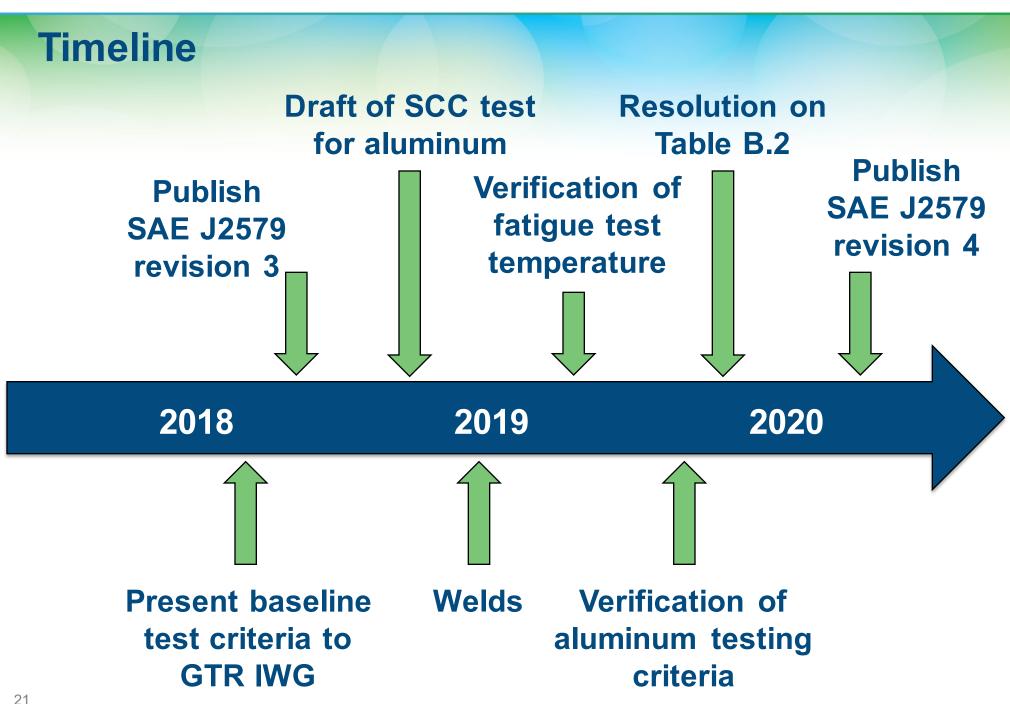
#### Strain-hardened austenitic stainless steel



- 1/3 Su of high-strength materials can be more then specified minimum yield strength of annealed material
- Implicitly, increase of design stress enables lower weight and lower cost designs without compromising performance
  - Justified by fatigue performance

### **Open questions**

- Temperature for fatigue life testing
  - Most data suggest that austenitic stainless steels show longer fatigue life at low temperature
  - Change temperature of fatigue test to room temperature only?
- Welding
  - Additional requirements?
- Additional testing requirements for aluminum alloys
  - Stress corrosion cracking (SCC) threshold
  - Test method and evaluation criteria for SCC being formulated by High-Pressure Institute of Japan HPIS E 103:2018
    - Method seems equivalent to ISO 7539-6
    - Criteria should be incorporated in SAE J2579
- How to incorporate "new" materials into SAE J2579
  - Replace table B.2 and periodically update with tested materials?



### **Summary**

- Materials compatibility test method in SAE J2579 provides performance-based metrics to evaluate materials for hydrogen service
  - J2579 Appendix B.3 requirements for materials do not purport to generate design data
  - Method consists for 4 parts
    - 1: Materials definition
    - 2: Slow strain rate tensile testing (3 tests)
    - 3: Fatigue life testing (3-5 tests)
    - 4: Evaluation of welds (if welded)
  - Tensile testing (SSRT) in H2 demonstrates that materials satisfy the specified minimum properties consistent with pressure application
  - Fatigue life testing in H2 demonstrates that materials have fatigue performance consistent with baseline materials

### **Backup slides**

# Fatigue life at low temperature appears to be greater than at room temperature R = 0.1, f = 1 Hz

800 800 Pressure Temperature 10 MPa: open 293K: open 103 MPa: closed 700 223K: closed 700 600 600 Maximum Stress (MPa) Maximum stress (MPa) 500 500 400 400 300 Temperature: 293K 300 Pressure: 10 MPa 316L (annealed) 316L (annealed) 200 316L (strain hardened) 200 316L (strain hardened) **XM-11 XM-11** Nitronic 60 Nitronic 60 100 SCF-260 100 SCF-260 304L 304L 10<sup>5</sup>  $10^2$ 10<sup>3</sup>  $10^{4}$ 10<sup>6</sup>  $10^{2}$  $10^{3}$  $10^{4}$  $10^{5}$ 10<sup>6</sup> Cycles to Failure Cycles to Failure

- Pressure has modest effect, if any, on fatigue life
- Temperature has either no effect or increases fatigue life
- Nitronic 60 is an exception for both pressure and temperature